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## AIRWORTHINESS DIRECTIVE REVIEW

ADR-D212-725

# *Eagle Single*

BELL 212 MODELS

Prepared By:

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Revision: **N**

Date: 15.07.28

**REVISION RECORD**

Revision	Issue Date	Description
A	10.06.01	New Issue
B	10.11.16	Add AD 2010-24-51
C	10.11.30	AD 2010-24-51 superseded; AD 2010-24-52 superseded; Add AD 2010-25-51
D	10.12.14	AD 2007-19-53 superseded; Add AD 2010-26-52
E	11.04.12	AD 2010-25-51 superseded; Add AD 2011-08-01
F	11.06.21	Add AD 2011-12-08
G	11.11.08	AD 2010-03-03 and AD 2007-19-53 superseded; add AD 2011-23-02
H	11.12.20	Add AD 2011-25-01
I	12.11.21	Add AD 2012-22-06
J	13.04.03	Add AD 2012-26-11, AD 2012-17-08, AD 2013-03-16, AD 2013-0065-E, AD 2013-0077-E and AD 2013-06-51
K	13.09.17	Add AD 2013-15-02 and AD 2013-18-07
L	13.11.26	Add AD EU 2013-0275
M	14.06.24	Add AD 2014-12-04, AD 2003-01-04 superseded
N	15.07.28	Add AD EU 2014-0201, AD 2015-15-04

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## 1. INTRODUCTION

### 1.1. Purpose

The purpose of this document is to provide information regarding Airworthiness Directives applicability to Bell 212 helicopters modified per TC STC SH07-28 (Eagle Single).

### 1.2. Review Process

The Transport Canada Website and FAA Website are reviewed on a weekly basis for new Airworthiness Directives that are applicable to Bell 212, 205A-1, and Honeywell T5317A/B Series engines. These Airworthiness Directives are reviewed and applicability is determined at that time.

### 1.3. Revision Process

When a new Airworthiness Directive is issued for the Bell 212, Bell 205A-1, Bell 205B, or Honeywell T5317A/B/BCV by either Transport Canada or the FAA, this document will be amended to show the new Airworthiness Directive and its applicability in Tables 1 – 4 in Section 2.

### 1.4. Distribution

This document and each subsequent revision will be made available to each Eagle Single operator for informational purposes at [www.dartaero.com](http://www.dartaero.com).

### 1.5. Use of This Document

Many of the Airworthiness Directives that apply to Eagle Single are applicable because the aircraft is a Bell 212 aircraft. However, the application of the Airworthiness Directive is no different than it would be on a standard Bell 212. In these situations, the "Affected by SH07-28" column of the table is designated as "No". Therefore, operators need to pay particular attention to Airworthiness Directives that appear as "Yes" in the "Affected by SH07-28" column of the tables. These situations are highlighted and special notes are provided to explain how SH07-28 addresses the Airworthiness Directive requirement. Special notes are also provided when existing Bell 212 Airworthiness Directives do not apply to SH07-28.

2. AIRWORTHINESS DIRECTIVE REVIEW

2.1. Bell 212

TABLE 1 – AIRWORTHINESS DIRECTIVES APPLICABLE TO BELL 212 HELICOPTERS

MODEL	COUNTRY	AD NUMBER	AD SUBJECT	APPLIES TO EAGLE SINGLE	AFFECTED BY SH07-28
212	CF	CF-98-15	External Rescue Systems	Yes	No
212	CF	CF-97-04	Fuel Line - Change To Stainless Elbow Fitting	No <sup>1</sup>	N/A
212	CF	CF-96-06	Emergency Escape Panels	No <sup>2</sup>	N/A
212	CF	CF-95-20R1	Boits Of Flight Control Servo Actuators	Yes	No
212	CF	CF-94-07	Zinc Primer As Overheat Indicator	Yes	No
212	CF	CF-91-29R1	Aft Facing Seat Headrests	Yes	No
212	CF	CF-91-27	Emergency Escape Release Handle Covers	No <sup>3</sup>	N/A
212	CF	CF-90-16R3	Main Rotor Yoke	Yes	No
212	CF	CF-88-23R1	Blade Strap Pins	Yes	No
212	EU	2014-0201	Superseded in Canada by FAA AD 2013-06-51	Superseded	No
212	EU	2013-0275	Superseded in Canada by FAA AD 2013-06-51	Superseded	No
212	EU	2013-0077-E	Goodrich Hoist Inspection and Test	Yes	No
212	EU	2013-0065-E	Superseded by AD 2013-0077-E	Superseded	No
212	EU	2006-0241	Replacement Of Main Case P/N BC85-051 And Piston BC85-052 With Improved Units Made Of Stainless Steel 6S80D.	Yes	No
212	US	2015-15-04	Main Rotor Blade Grip Replacement	Yes	No
212	US	2014-12-04	Main Rotor Grip Inspection	Yes	No
212	US	2013-15-02	Fatigue cracking of Tail Rotor Blades	Yes	No
212	US	2013-06-51	Goodrich Hoist Inspection and Test	Yes	No
212	US	2013-03-16	Main Rotor Hub Inboard Strap Fitting - Inspection	Yes	No
212	US	2012-22-06	Aeronautical Accessories – High Landing Gear Forward Crosstube Inspection	Yes	No
212	US	2012-17-08	Main Rotor Yoke Inspection and new life limit	Yes	No
212	US	2011-25-01	Apical Industries, placard installation	Yes	No
212	US	2011-23-02	Main Rotor Blade detection of edge void corrosion or cracks	Yes	No
212	US	2011-12-08	Corrosion and pitting in forward spar area of T/R blade	Yes	No
212	US	2011-08-01	Superseded by AD 2013-03-16	Superseded	No

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TABLE 1 – AIRWORTHINESS DIRECTIVES APPLICABLE TO BELL 212 HELICOPTERS

MODEL	COUNTRY	AD NUMBER	AD SUBJECT	APPLIES TO EAGLE SINGLE	AFFECTED BY SH07-28
212	US	2010-26-52	Missing Adhesive to the tip weight screws	Yes	No
212	US	2010-25-51	Superseded by AD 2011-08-01	Superseded	No
212	US	2010-24-52	Superseded by AD 2010-25-51	Superseded	No
212	US	2010-24-51	Superseded by AD 2010-24-52	Superseded	No
212	US	2010-10-16	Replacement of Aeronautical Accessories Low Skid Landing Gear Forward Crosstubes	Yes	No
212	US	2010-03-03	Superseded by AD 2011-23-02	Superseded	No
212	US	2008-10-03	Superseded by AD 2013-15-02	Superseded	No
212	US	2007-22-02	Tail Rotor Blade (Blade) Forward Tip Weight Retention Block (Tip Block) And The Aft Tip Closure (Tip Closure)	Yes	No
212	US	2007-19-53	Superseded by AD 2010-26-52	Superseded	No
212	US	2005-20-38	To Prevent Rupture Of An Adapter, Uncontrolled Jetting Of Pressurized Gas From The Nitrogen Bottle	Yes	No
212	US	2003-01-04	Superseded by AD 2014-12-04	Superseded	No
212	US	2002-22-14	204B, 205A/B, 212, 214B/B1 - Main Rotor TT Straps	Yes	No
212	US	2002-19-05	212 Models - Vertical Fin Forward Spar Cap	Yes	No
212	US	2002-09-04	Superseded by 2007-22-02	Superseded	No
212	US	2001-13-01	Tail Rotor Counterweight Bellcrank Retention Nut	Yes	No
212	US	2001-08-04	Main Rotor Actuator Locking Washer	Yes	No
212	US	2000-15-52	Main Rotor Mast/Trunnion	Yes	No
212	US	98-11-15	Tail Rotor Yoke Assy Failure.	Yes	No
212	US	94-18-09	Main Transmission Lower Planetary Spider	Yes	No
212	US	93-17-12	Main Rotor Transmission Lower Planetary Spider	Yes	No
212	US	93-05-01	Superseded by AD 2012-17-08	Superseded	No
212	US	92-23-01	Prevent Separation Main Rotor Pillow Blocks	Yes	No
212	US	92-13-10	Tail Rotor Driveshaft Hanger Bearings	Yes	No
212	US	92-09-05	Eng Driveshaft Coupling Adapter	No <sup>4</sup>	N/A
212	US	92-07-08	Swashplate Support Assembly	Yes	No
212	US	90-26-11	Main Rotor Drag Brace Assembly	Yes	No
212	US	90-03-09	T/R Trunnion Bearing Housing Assembly	Yes	No
212	US	89-08-05	Main Transmission - Spiral Bevel Gears	Yes	No
212	US	86-17-09	Tail Rotor Trunnion Bearing Housing	Yes	No
212	US	81-19-01	Superseded by AD 2012-17-08	Superseded	No
212	US	81-10-07	Swashplate Support	Yes	No
212	US	80-21-05	Landing Gear Cross Tubes	Yes	No
212	US	79-20-05	Superseded by AD 2012-17-08	Superseded	No

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TABLE 1 – AIRWORTHINESS DIRECTIVES APPLICABLE TO BELL 212 HELICOPTERS

MODEL	COUNTRY	AD NUMBER	AD SUBJECT	APPLIES TO EAGLE SINGLE	AFFECTED BY SH07-28
212	US	79-05-08	Emergency Flotation (Ditching) Equipment	Yes	No
212	US	78-21-02	External Load Link Assembly	Yes	No
212	US	78-20-07	Shoulder Radius Fitting	Yes	No
212	US	78-17-03	Spiral Bevel Gear	Yes	No
212	US	78-14-07	Skid Landing Gear	Yes	No
212	US	78-09-02	Emergency Float Kits	Yes	No
212	US	77-17-05	Emergency Exit Latch Pin	No <sup>5</sup>	N/A
212	US	77-17-03	Tail Rotor Blade Pitch Horn	Yes	No
212	US	77-10-05	Cracks In Main Rotor Blades Skin	Yes	No
212	US	77-05-02	Main Rotor Blades	Yes	No
212	US	76-14-03	Cross Tube Assemblies	Yes	No
212	US	76-02-06	Main Rotor Blades	Yes	No
212	US	75-26-05	Main Rotor Blades	Yes	No
212	US	75-26-03	Emergency Exit Handle	No <sup>6</sup>	N/A
212	US	75-07-01	High Frequency Vibration	Yes	No
212	US	74-20-05	Elevator Modification	Yes	No
212	US	74-02-01	Tail Rotor Trunnion Bearings	Yes	No

- <sup>1</sup> AD CF-97-04 not applicable to Eagle Single because PT6 engines no longer installed on aircraft
- <sup>2</sup> AD CF-96-06 not applicable to Eagle Single because escape panels have been modified per SH04-31
- <sup>3</sup> AD CF-91-27 not applicable to Eagle Single because escape panels have been modified per SH04-31
- <sup>4</sup> AD 92-09-05 not applicable to Eagle Single because PT6 engines no longer installed on aircraft
- <sup>5</sup> AD 77-17-05 not applicable to Eagle Single because escape panels have been modified per SH04-31
- <sup>6</sup> AD 75-26-03 not applicable to Eagle Single because escape panels have been modified per SH04-31

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## 2.2. Pratt and Whitney PT6T

None of the Airworthiness Directives that apply to Pratt and Whitney PT6T engines are applicable to the Bell 212 with TC STC SH07-28 embodied since these engines are no longer installed on the aircraft.

## 2.3. Honeywell T5317A

TABLE 2 – AIRWORTHINESS DIRECTIVES APPLICABLE TO HONEYWELL T5317A ENGINES					
MODEL	COUNTRY	AD NUMBER	AD SUBJECT	APPLIES TO EAGLE SINGLE	AFFECTED BY SH07-28
T5317A	US	2009-15-13	Engine Combustion Chamber Housing (CCH) - Detection Of Cracks.	Yes	No
T5317A	US	2006-11-16	Engine Fuel Control Regulator Assembly Failure Of Both Main And Secondary Drive Shaft And Pump Gear Splines.	Yes	No
T5317A	US	2006-01-05	Service Life Limits	Yes	No
T5317A	US	77-17-08	T53 Series - Fuel Regulator P1 Multiplier Lever	Yes	No
T5317A	US	2002-15-04	T5313B, T5317, + T53 - Compressor Life Limits	Yes	No
T5317A	US	2002-09-09	Centrifugal Compressor Impeller Prorate/Inspection	Yes	No
T5317A	US	2002-03-01	Failure Of The Tachometer Drive Spur Gear	Yes	No
T5317A	US	97-07-05	N2 Spur Gear Nut Retainer	Yes	No
T5317A	US	97-21-07R1	Accessory Drive Carrier Assembly	Yes	No
T5317A	US	95-10-04	Centrifugal Compressor Impeller	Yes	No
T5317A	US	75-22-02	Torquemeter Boost Pump Drive Shaft	Yes	No

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2.4. Honeywell T5317B

TABLE 3 – AIRWORTHINESS DIRECTIVES APPLICABLE TO HONEYWELL T5317B ENGINES					
MODEL	COUNTRY	AD NUMBER	AD SUBJECT	APPLIES TO EAGLE SINGLE	AFFECTED BY SH07-28
T5317B	US	2009-15-13	Engine Combustion Chamber Housing (CCH) - Detection Of Cracks.	Yes	No
T5317B	US	2006-11-16	Engine Fuel Control Regulator Assembly Failure Of Both Main And Secondary Drive Shaft And Pump Gear Splines.	Yes	No
T5317B	US	2006-01-05	Service Life Limits	Yes	No
T5317B	US	77-17-08	T53 Series - Fuel Regulator P1 Multiplier Lever	Yes	No
T5317B	US	2002-15-04	T5313B, T5317, + T53 - Compressor Life Limits	Yes	No
T5317B	US	2002-03-01	Failure Of The Tachometer Drive Spur Gear	Yes	No
T5317B	US	97-07-05	N2 Spur Gear Nut Retainer	Yes	No
T5317B	US	97-21-07R1	Accessory Drive Carrier Assembly	Yes	No
T5317B	US	95-10-04	Centrifugal Compressor Impeller	Yes	No
T5317B	US	75-22-02	Torquemeter Boost Pump Drive Shaft	Yes	No

2.5. Honeywell T5317BCV

TABLE 3 – AIRWORTHINESS DIRECTIVES APPLICABLE TO HONEYWELL T5317BCV ENGINES					
MODEL	COUNTRY	AD NUMBER	AD SUBJECT	APPLIES TO EAGLE SINGLE	AFFECTED BY SH07-28
T5317BCV	US	2009-15-13	Engine Combustion Chamber Housing (CCH) - Detection Of Cracks.	Yes	No

2.6. Bell 205A-1

TABLE 4 – AIRWORTHINESS DIRECTIVES FOR BELL 205A-1 HELICOPTERS					
MODEL	COUNTRY	AD NUMBER	AD SUBJECT	APPLIES TO EAGLE SINGLE	AFFECTED BY SH07-28
205A-1	CF	CF-98-15	External Rescue Systems	See Bell 212 AD List Table 1	
205A-1	CF	CF-96-06	Emergency Escape Panels	See Bell 212 AD List Table 1	
205A-1	CF	CF-95-20R1	Bolts Of Flight Control Servo Actuators	See Bell 212 AD List Table 1	
205A-1	CF	CF-94-07	Zinc Primer As Overheat Indicator	See Bell 212 AD List Table 1	
205A-1	CF	CF-91-29R1	Aft Facing Seat Headrests	See Bell 212 AD List Table 1	
205A-1	CF	CF-91-27	Emergency Escape Release Handle Covers	See Bell 212 AD List Table 1	
205A-1	CF	CF-90-16R3	Main Rotor Yoke	See Bell 212 AD List Table 1	
205A-1	CF	CF-88-23R1	Blade Strap Pins	See Bell 212 AD List Table 1	
205A-1	CF	CF-77-03	205A - Alignment Between Engine And Transmission	Yes <sup>7</sup>	No
205A-1	US	2015-15-04	Main Rotor Blade Grip Replacement	See Bell 212 AD List Table 1	
205A-1	US	2014-12-04	Main Rotor Grip Inspection	See Bell 212 AD List Table 1	
205A-1	US	2013-18-07	Tail Rotor Pitch Control Chain Inspection and Replacement	No	N/A
205A-1	US	2013-15-02	Fatigue cracking of Tail Rotor Blades	See Bell 212 AD List Table 1	
205A-1	US	2013-03-16	Main Rotor Hub Inboard Strap Fitting - Inspection	See Bell 212 AD List Table 1	
205A-1	US	2012-26-11	Starter/Generator Power Cable Replacement	No	N/A
205A-1	US	2012-22-06	Aeronautical Accessories – High Landing Gear Forward Crosstube - Inspection	See Bell 212 AD List Table 1	
205A-1	US	2012-17-08	Main Rotor Yoke Inspection and new life limit	See Bell 212 AD List Table 1	
205A-1	US	2011-23-02	Main Rotor Blade detection of edge void, corrosion and cracks	See Bell 212 AD List Table 1	
205A-1	US	2011-12-08	Corrosion and pitting in forward spar area of T/R blade	See Bell 212 AD List Table 1	
205A-1	US	2010-26-52	Missing adhesive to the tip weight screws	See Bell 212 AD List Table 1	
205A-1	US	2010-10-16	Replacement of Aeronautical Accessories Low Skid Landing Gear Forward Crosstubes	See Bell 212 AD List Table 1	
205A-1	US	2008-10-03	Inspection Of Specific Tail Rotor Blades.	See Bell 212 AD List Table 1	
205A-1	US	2007-22-02	Tail Rotor Blade (Blade) Forward Tip Weight Retention Block (Tip Block) And The Aft Tip Closure (Tip Closure)	See Bell 212 AD List Table 1	
205A-1	US	2007-19-53	Superseded by AD 2010-26-52	See Bell 212 AD List Table 1	
205A-1	US	2003-01-04	Superseded by AD 2014-12-04	See Bell 212 AD List Table 1	
205A-1	US	2002-22-14	204B, 205A/B, 212, 214B/B1 - Main Rotor TT Straps	See Bell 212 AD List Table 1	
205A-1	US	2002-09-51	Tail Rotor Grips	No	N/A

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TABLE 4 – AIRWORTHINESS DIRECTIVES FOR BELL 205A-1 HELICOPTERS

MODEL	COUNTRY	AD NUMBER	AD SUBJECT	APPLIES TO EAGLE SINGLE	AFFECTED BY SH07-28
205A-1	US	2001-13-01	Tail Rotor Counterweight Bellcrank Retention Nut	See Bell 212 AD List Table 1	
205A-1	US	2001-08-04	Main Rotor Actuator Locking Washer	See Bell 212 AD List Table 1	
205A-1	US	2000-15-52	Main Rotor Mast/Trunnion	See Bell 212 AD List Table 1	
205A-1	US	99-18-02	Vertical Fin Spar Cap	No	N/A
205A-1	US	99-17-03	Failure Of Tailboom Vertical Fin Spar	No	N/A
205A-1	US	98-11-14	Tail Rotor Yoke Assy Failure	No	N/A
205A-1	US	95-10-07	Tail Rotor Drive Gearbox Assembly	No	N/A
205A-1	US	94-18-09	Main Transmission Lower Planetary Spider	See Bell 212 AD List Table 1	
205A-1	US	93-17-12	Main Rotor Transmission Lower Planetary Spider	See Bell 212 AD List Table 1	
205A-1	US	92-27-21	Tail Rotor Drive Shaft	Yes <sup>9</sup>	No
205A-1	US	92-23-01	Prevent Separation Main Rotor Pillow Blocks	See Bell 212 AD List Table 1	
205A-1	US	92-13-10	Tail Rotor Driveshaft Hanger Bearings	See Bell 212 AD List Table 1	
205A-1	US	92-07-08	Swashplate Support Assembly	See Bell 212 AD List Table 1	
205A-1	US	90-03-10	Tail Rotor Grips	No	N/A
205A-1	US	90-03-09	Tail Rotor Hub Assembly	See Bell 212 AD List Table 1	
205A-1	US	89-20-12	Tail Rotor Hub Assembly	No	N/A
205A-1	US	89-08-05	Transmission Internal Sump Oil Filter	See Bell 212 AD List Table 1	
205A-1	US	88-25-05	Tail Rotor Grip Assembly	No	N/A
205A-1	US	86-17-10	Tail Rotor Trunnion Bearing Housing	No	N/A
205A-1	US	81-19-02	Superseded by AD 2012-17-08	See Bell 212 AD List Table 1	
205A-1	US	80-21-05	Landing Gear Cross Tubes	See Bell 212 AD List Table 1	
205A-1	US	80-03-07	Fuselage Main Beam Splices	No	No
205A-1	US	79-20-05	Superseded by AD 2012-17-08	See Bell 212 AD List Table 1	
205A-1	US	78-21-02	External Load Link Assembly	See Bell 212 AD List Table 1	
205A-1	US	78-20-07	Shoulder Radius Fitting	See Bell 212 AD List Table 1	
205A-1	US	78-14-07	Skid Landing Gear	See Bell 212 AD List Table 1	
205A-1	US	77-17-05	Emergency Exit Latch Pin	See Bell 212 AD List Table 1	
205A-1	US	77-17-03	Tail Rotor Blade Pitch Horn	See Bell 212 AD List Table 1	
205A-1	US	77-10-07	Engine-To-Transmission Assy	Yes <sup>9</sup>	No
205A-1	US	76-14-03	Cross Tube Assemblies	See Bell 212 AD List Table 1	
205A-1	US	76-12-07	Superseded by AD 2013-18-07	See Bell 212 AD List Table 1	
205A-1	US	76-10-01	Tail Boom	No	N/A
205A-1	US	76-06-02	Fire Extinguisher Circuit	No	N/A
205A-1	US	76-02-06	Main Rotor Blades	Yes <sup>10</sup>	No
205A-1	US	75-26-05	Main Rotor Blades	See Bell 212 AD List Table 1	
205A-1	US	75-26-03	Emergency Exit Handle	See Bell 212 AD List Table 1	
205A-1	US	74-23-02	Tail Rotor Pitch	No	N/A
205A-1	US	74-02-01	Tail Rotor Trunnion Bearings	See Bell 212 AD List Table 1	
205A-1	US	73-17-04	Tail Rotor Grips	No	N/A
205A-1	US	73-16-03	Landing Gear Cross Tubes	No	N/A
205A-1	US	71-21-02	Tail Fin And Boom	No	N/A
205A-1	US	70-06-02	Tail Rotor Grip Assy	No	N/A
205A-1	US	69-15-07	Float Kit Tube	No	N/A

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<sup>7</sup> AD CF-77-03 applies to Eagle Single due to the engine/transmission installation. ICA-D212-725 contains the updated information to comply with the requirements of this AD.

<sup>8</sup> AD 92-27-21 applies to Eagle Single. The D205-765-011 Driveshaft Tunnel Kit installed per TCCA SH08-12 (ref. FAA STC SR02552NY) does not alleviate the initial and recurring inspection requirements of this AD.

<sup>9</sup> AD 77-10-07 applies to Eagle Single due to the engine/transmission installation. ICA-D212-725 contains the updated information to comply with the requirements of this AD.

<sup>10</sup> AD 76-06-02 applies to Eagle Single because SH07-28 uses the Bell 205A-1 engine fire extinguishing system.

2.7. Bell 205B

TABLE 5 – AIRWORTHINESS DIRECTIVES FOR BELL 205B HELICOPTERS

MODEL	COUNTRY	AD NUMBER	AD SUBJECT	APPLIES TO EAGLE SINGLE	AFFECTED BY SH07-28
205B	CF	CF-98-15	External Rescue Systems	See Bell 212 AD List Table 1	
205B	CF	CF-88-23R1	Blade Strap Pins	See Bell 212 AD List Table 1	
205B	US	2014-12-04	Main Rotor Grip Inspection	See Bell 212 AD List Table 1	
205B	US	2013-15-02	Fatigue cracking of Tail Rotor Blades	See Bell 212 AD List Table 1	
205B	US	2013-03-16	Main Rotor Hub Inboard Strap Fitting - Inspection	See Bell 212 AD List Table 1	
205B	US	2012-26-11	Starter/Generator Power Cable Replacement	See Bell 205A-1 AD List Table 4	
205B	US	2012-17-08	Main Rotor Yoke Inspection and new life limit	See Bell 212 AD List Table 1	
205B	US	2012-22-06	Aeronautical Accessories – High Landing Gear Forward Crosstube - Inspection	See Bell 212 AD List Table 1	
205B	US	2011-23-02	Main rotor blade detection of edge void, corrosion and cracks	See Bell 212 AD List Table 1	
205B	US	2011-12-08	Corrosion and pitting in forward spar area of T/R blade	See Bell 212 AD List Table 1	
205B	US	2010-26-52	Missing adhesive to the tip weight screws	See Bell 212 AD List Table 1	
205B	US	2010-10-16	Replacement of Aeronautical Accessories Low Skid Landing Gear Forward Crosstubes	See Bell 212 AD List Table 1	
205B	US	2010-03-03	Superseded by AD 2011-23-02	See Bell 212 AD list Table 1	
205B	US	2008-10-03	Inspection Of Specific Tail Rotor Blades.	See Bell 212 AD List Table 1	
205B	US	2007-22-02	Tail Rotor Blade (Blade) Forward Tip Weight Retention Block (Tip Block) And The Aft Tip Closure (Tip Closure)	See Bell 212 AD List Table 1	
205B	US	2007-19-53	Superseded by AD 2010-26-52	See Bell 212 AD List Table 1	
205B	US	2003-01-04	Superseded by AD 2014-12-04	See Bell 212 AD List Table 1	
205B	US	2002-22-14	204B, 205A/B, 212, 214B/B1 – Main Rotor TT Straps	See Bell 212 AD List Table 1	
205B	US	2002-09-51	Tail Rotor Grip	See Bell 205A-1 AD List Table 4	
205B	US	2001-13-01	Tail Rotor Counterweight Bellcrank Retention Nut	See Bell 212 AD List Table 1	
205B	US	2001-08-04	Main Rotor Actuator Locking Washer	See Bell 212 AD List Table 1	
205B	US	2000-15-52	Main Rotor Mast/Trunnion	See Bell 212 AD List Table 1	
205B	US	99-18-02	Vertical Fin Spar Cap	See Bell 205A-1 AD List Table 4	
205B	US	98-11-14	Tail Rotor Yoke Assembly	See Bell 205A-1 AD List Table 4	
205B	US	94-18-09	Main Transmission Lower Planetary Spider	See Bell 212 AD List Table 1	
205B	US	93-17-12	Main Rotor Transmission Lower Planetary Spider	See Bell 212 AD List Table 1	
205B	US	92-27-21	Tail Rotor Driveshaft	See Bell 205A-1 AD List Table 4	
205B	US	92-23-01	Prevent Separation Main Rotor Pillow Blocks	See Bell 212 AD List Table 1	

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TABLE 5 – AIRWORTHINESS DIRECTIVES FOR BELL 205B HELICOPTERS					
MODEL	COUNTRY	AD NUMBER	AD SUBJECT	APPLIES TO EAGLE SINGLE	AFFECTED BY SH07-28
205B	US	92-13-10	Tail Rotor Driveshaft Hanger Bearings	See Bell 212 AD List Table 1	
205B	US	92-07-08	Swashplate Support Assembly	See Bell 212 AD List Table 1	
205B	US	86-17-10	Tail Rotor Hub Assembly	See Bell 205A-1 AD List Table 4	
205B	US	81-19-02	Main Rotor Yoke	See Bell 205A-1 AD List Table 4	

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